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Australian Government

Department of Defence

Defence Science and
Technology Organisation

Preliminary Probabilistic Risk Analysis of a Plate with Multiple Open Holes Considering Multi-Site Damage

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Presented by Ribelito Torregosa

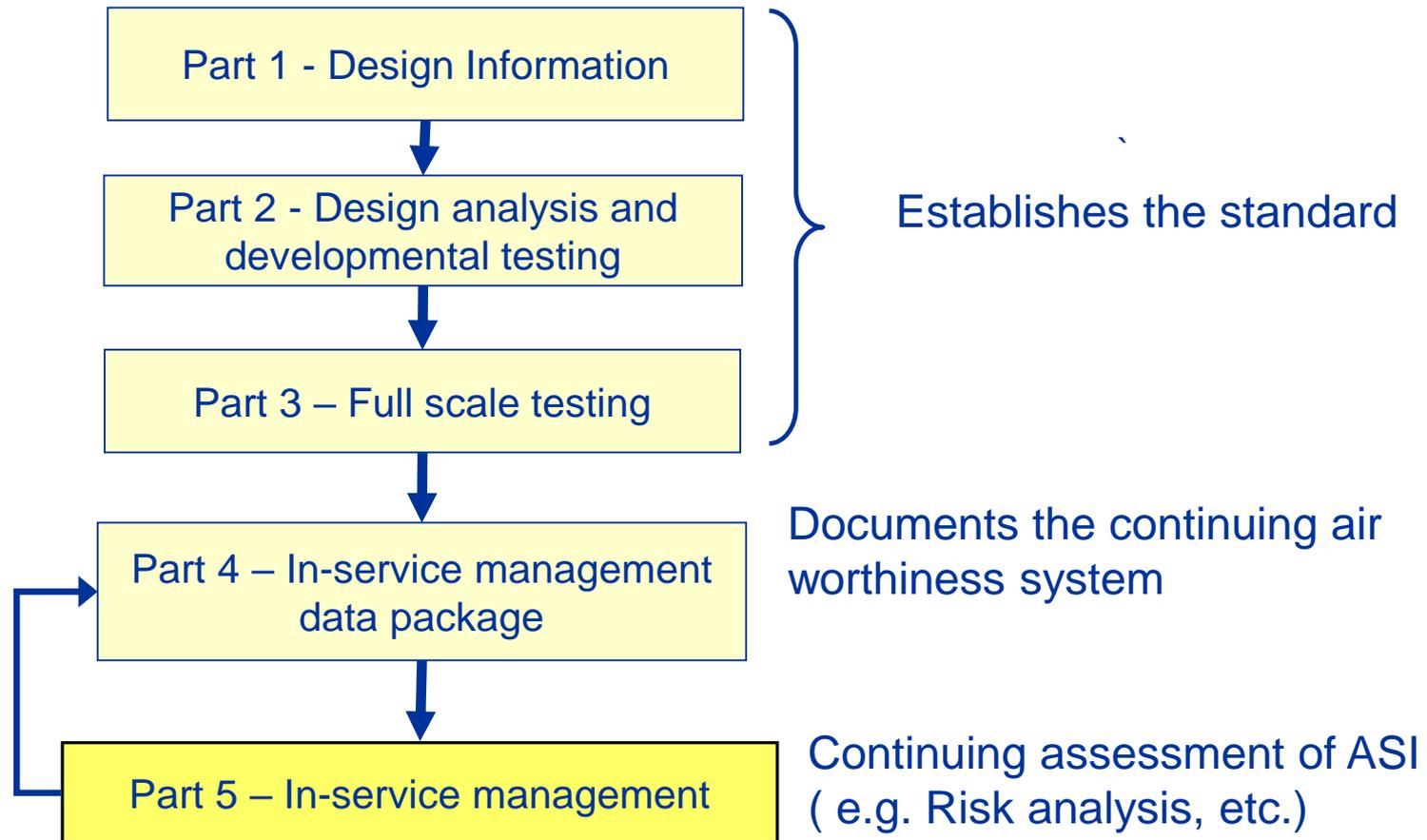
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Outline of the presentation

- Relevance of Probabilistic Risk Analysis (PRA) to aircraft structural integrity assessment and management of military aircraft
- Conventional phase by phase (PBP) risk analysis
- Multi-site damage (MSD) risk analysis
- Comparison of results between different MSD scenarios
- Conclusion



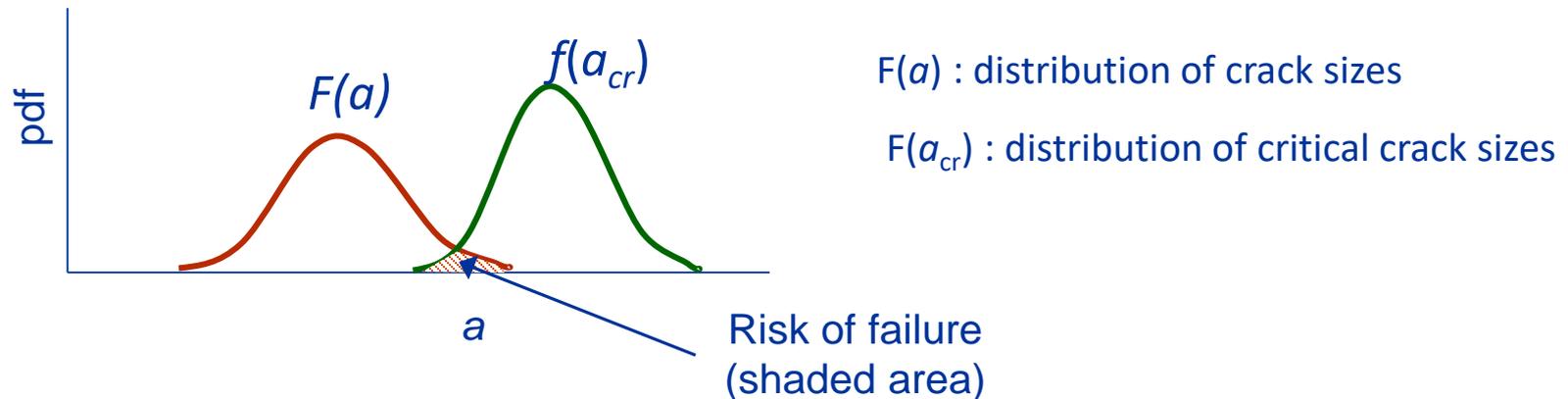
Role of probabilistic risk analysis in ASIP



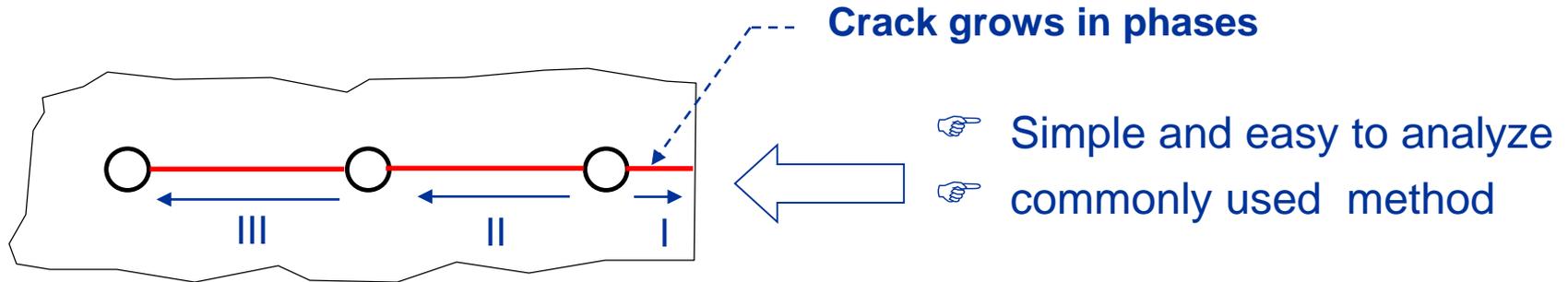
Aircraft Structural Integrity Program (ASIP) Parts

Definition of Risk

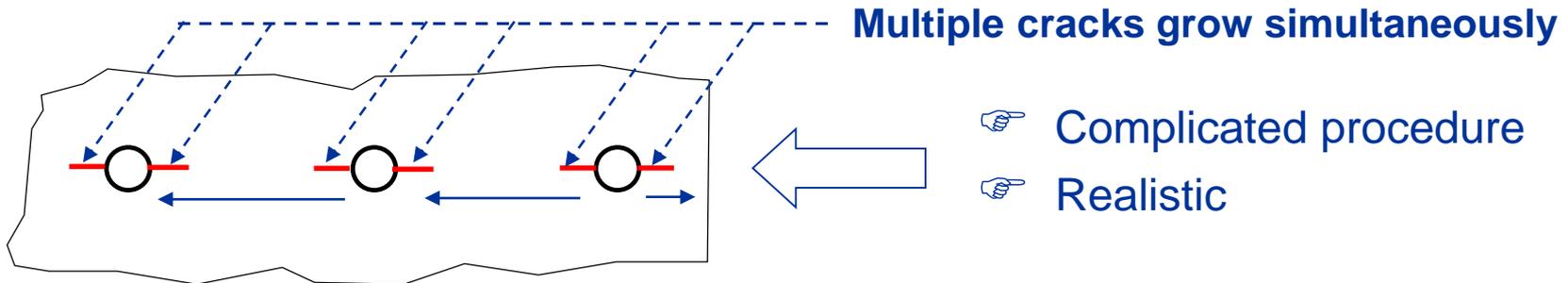
- For a component that contains a crack, the risk is the probability that the component will fail due to sudden fracture during the next flight.
- If the component is a key structure, then its failure also signals the failure of the aircraft
- The probability that a crack size, a exceeds the critical crack size, a_{cr} as described in the figure below.



Risk analysis methods



1) Phase by phase approach



2) Multisite damage (MSD)

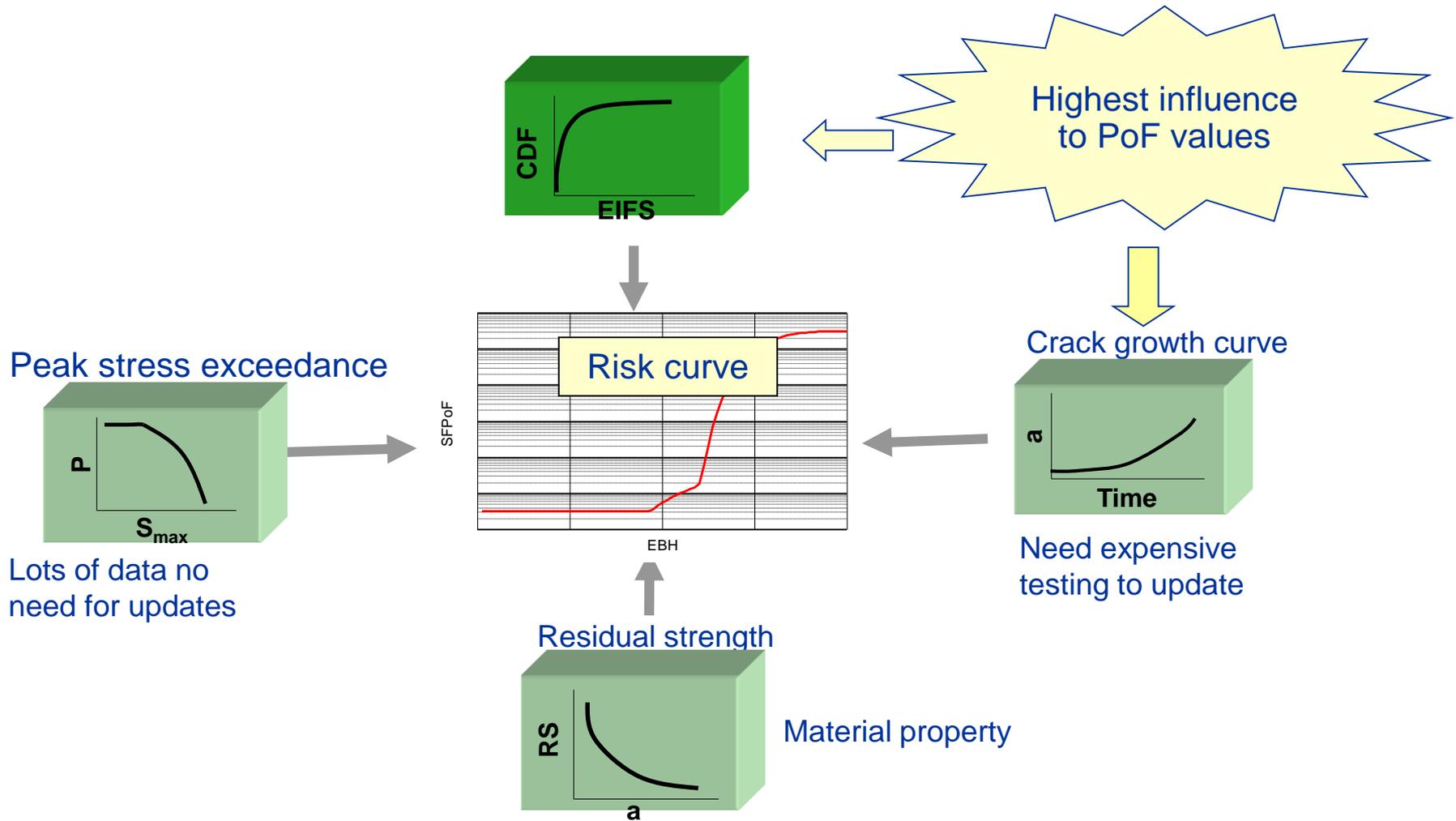
Focus of this presentation

Risk analysis methods (cont.)

- At least four groups of data are required for risk analysis
 - ✓ The distribution of the initial cracks
 - ✓ A crack growth curve from DADTA analysis
 - ✓ The exceedance distribution of the service load
 - ✓ The residual strength distribution of the crack configuration

- From experience, the data of most influence are
 - ✓ The distribution of the initial cracks
 - ✓ The crack growth curve from DADTA analysis

Graphical Representation of the Input Data



Why MSD analysis is important?

Maintaining structural integrity

“**multiple-site damage** could cause many small cracks in the structure, which grow slowly by themselves, to join one another over time, creating a much larger crack, and significantly reducing the expected time until failure”

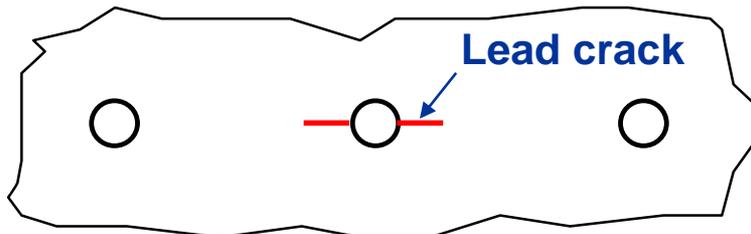
Required by standard

“Durability criteria apply to all airframe structural components and shall include criteria that pertain to the onset of **widespread fatigue damage**”

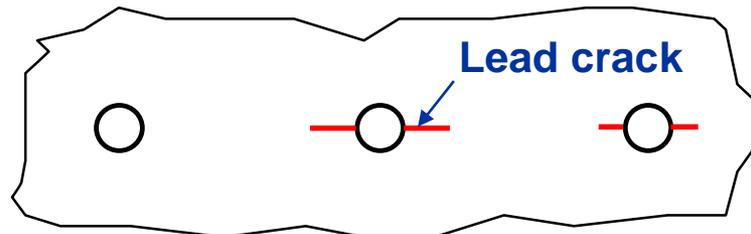
 MIL-STD-1530C (USAF) 5.1.3.4



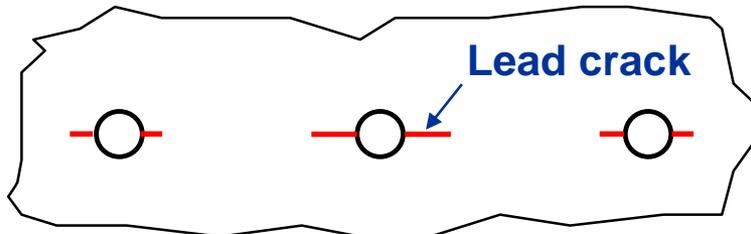
Effect of multiple cracks to the residual strength



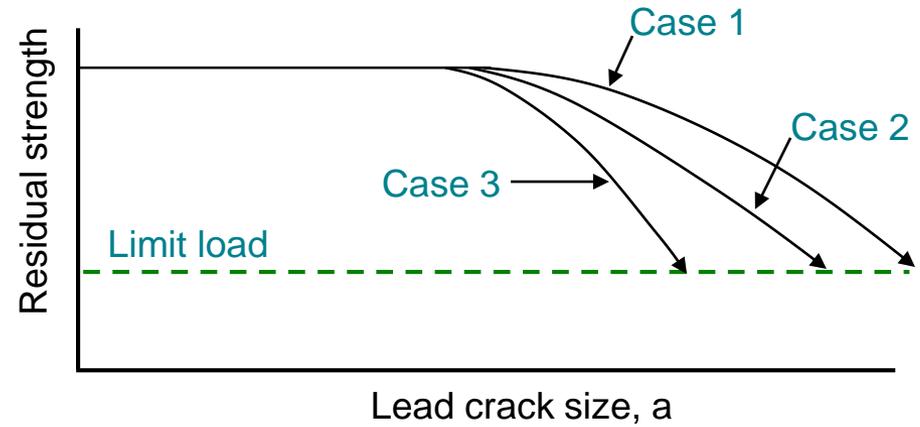
Case 1- Single crack



Case 2- MSD 2 cracks

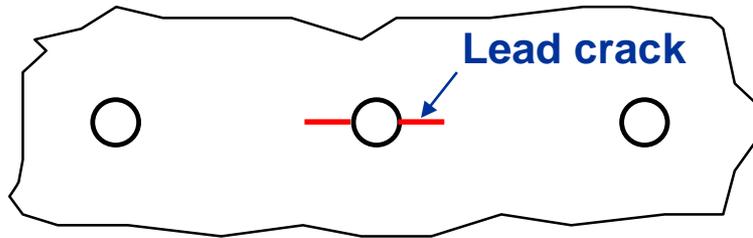


Case 3- MSD 3 cracks

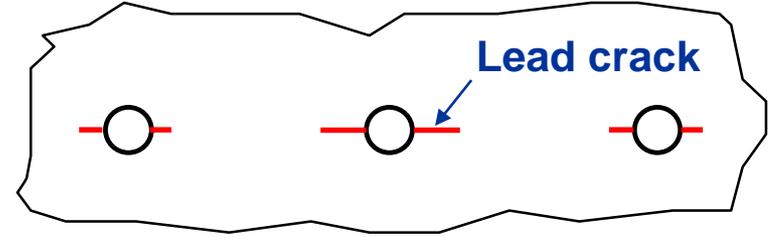


MSD reduces the residual strength

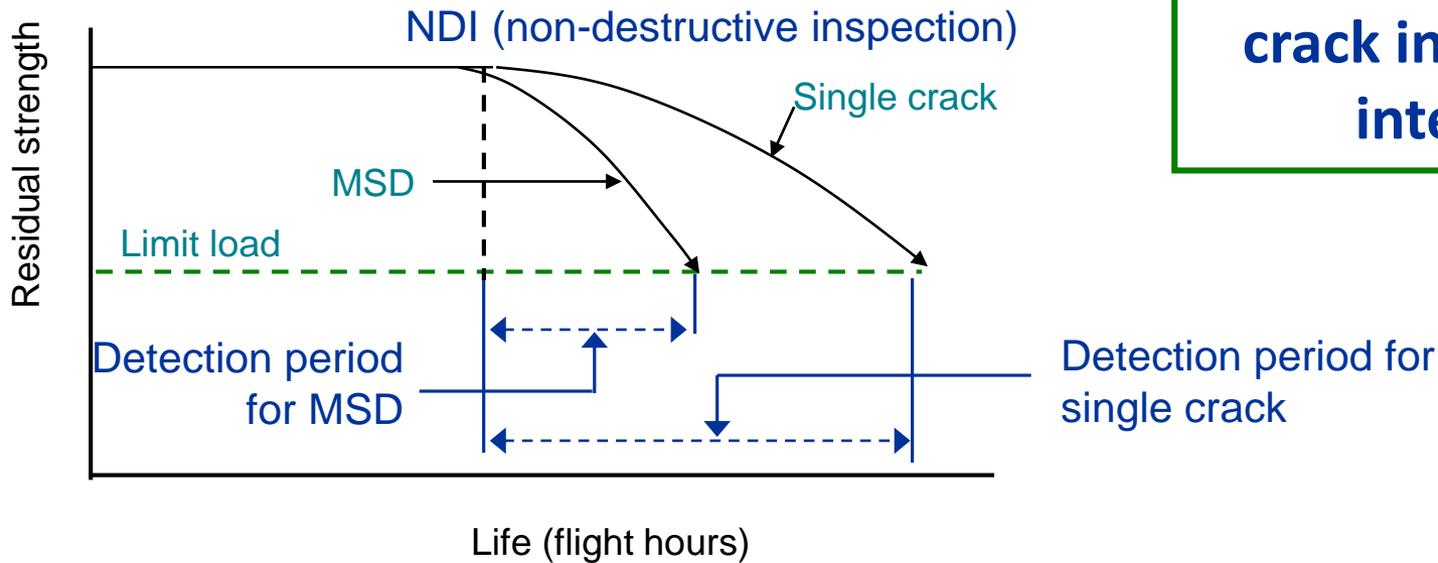
Effect of multiple cracks to the residual strength



Single crack

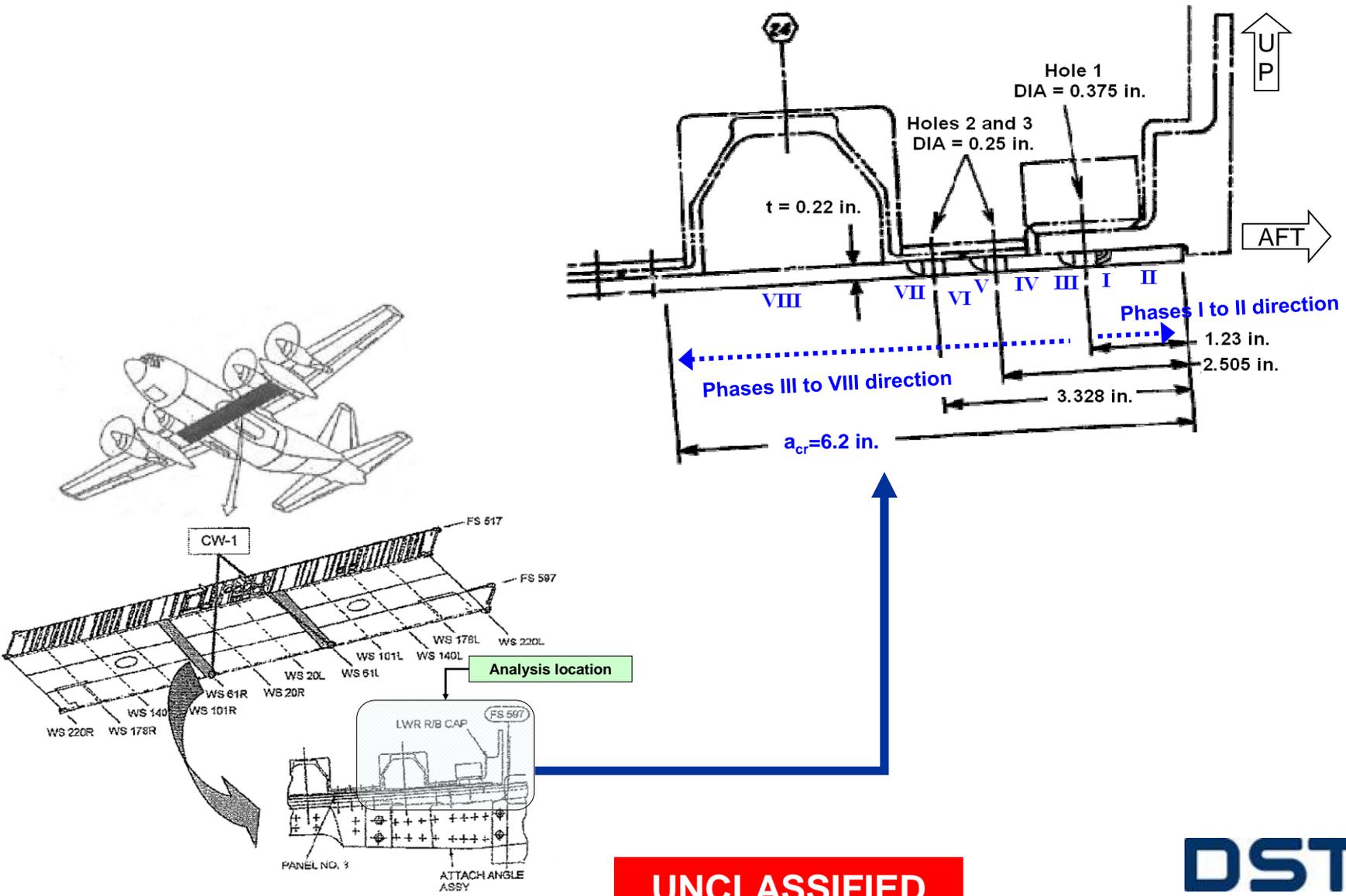


MSD

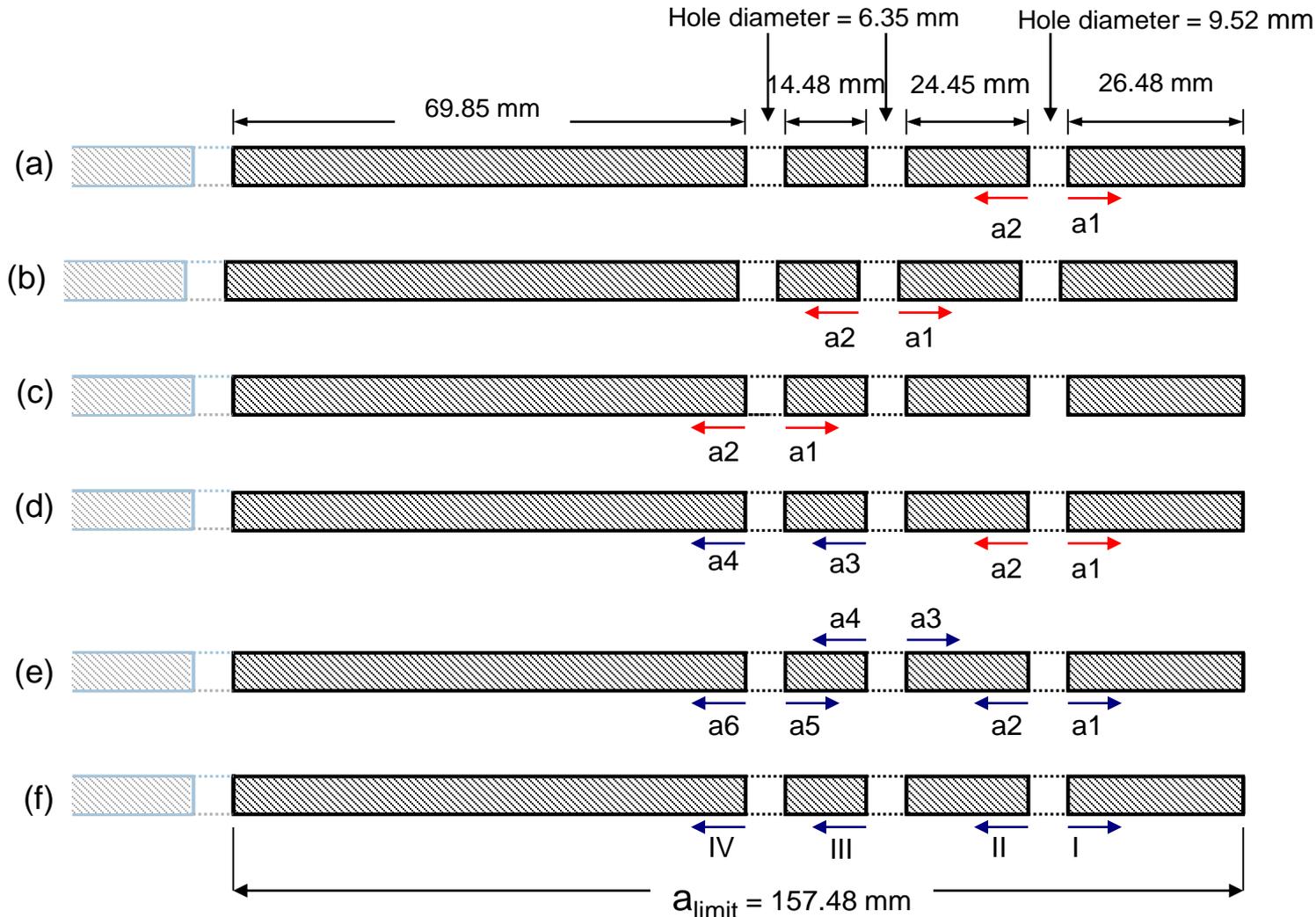


MSD reduces the crack inspection interval

Probabilistic Risk Analysis of C130-H CW-1 Location



C130-H CW-1 MSD crack scenarios analysed

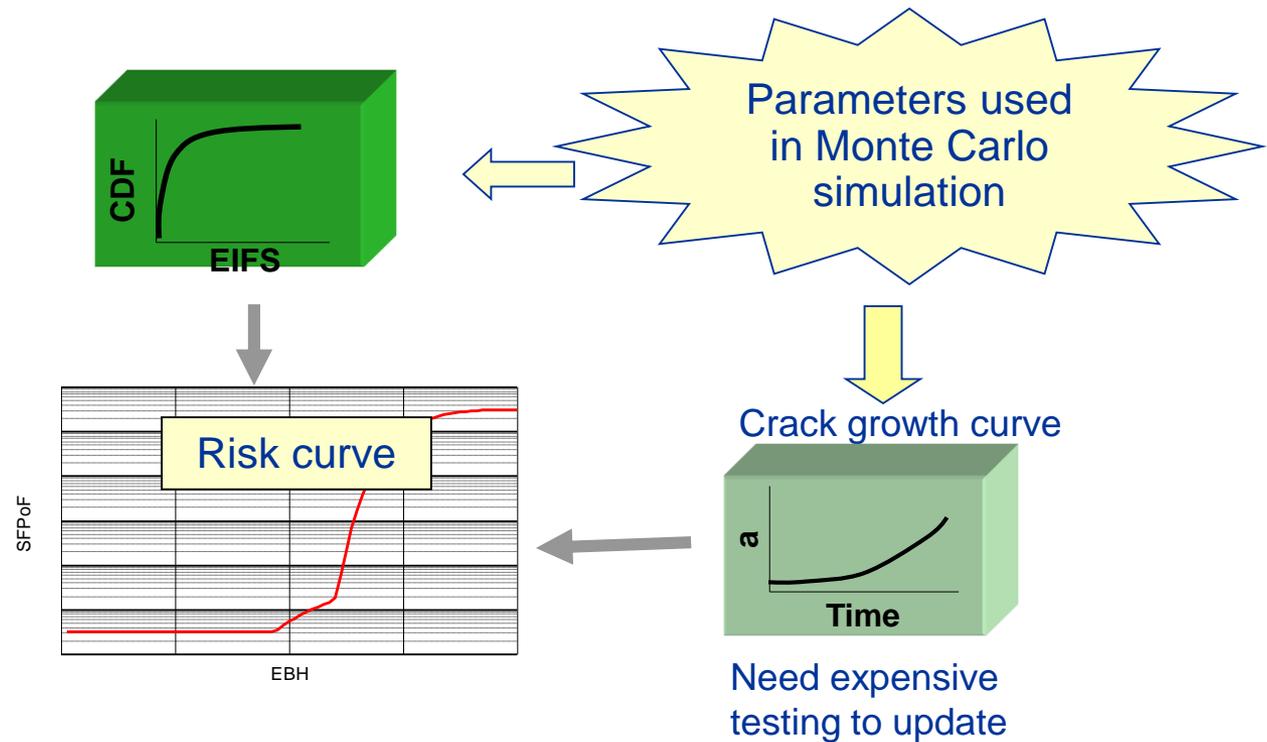


Note: Dimension not to scale

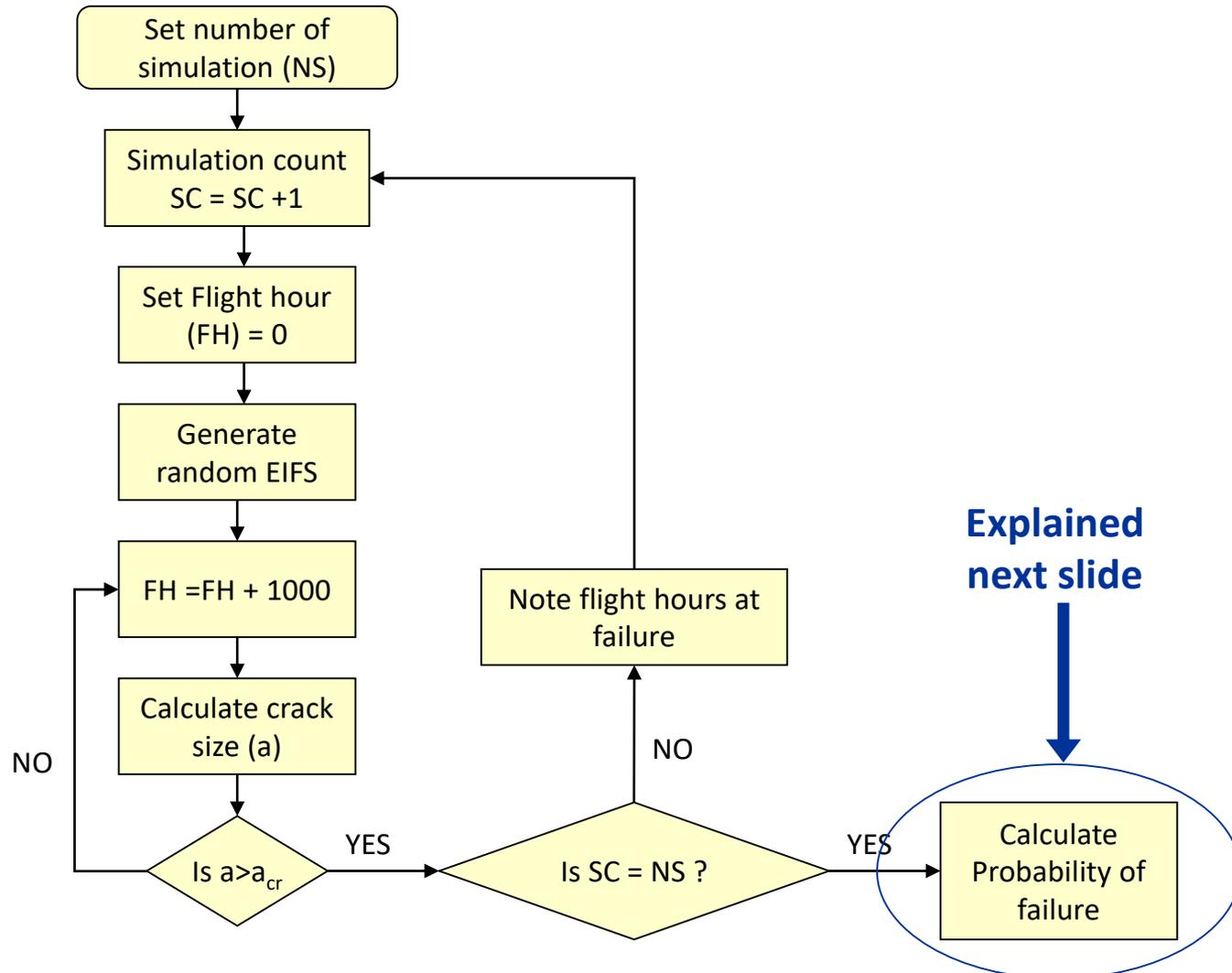
Legend:

← →
Lead crack

2 – Parameter Monte Carlo simulation of MSD Risks



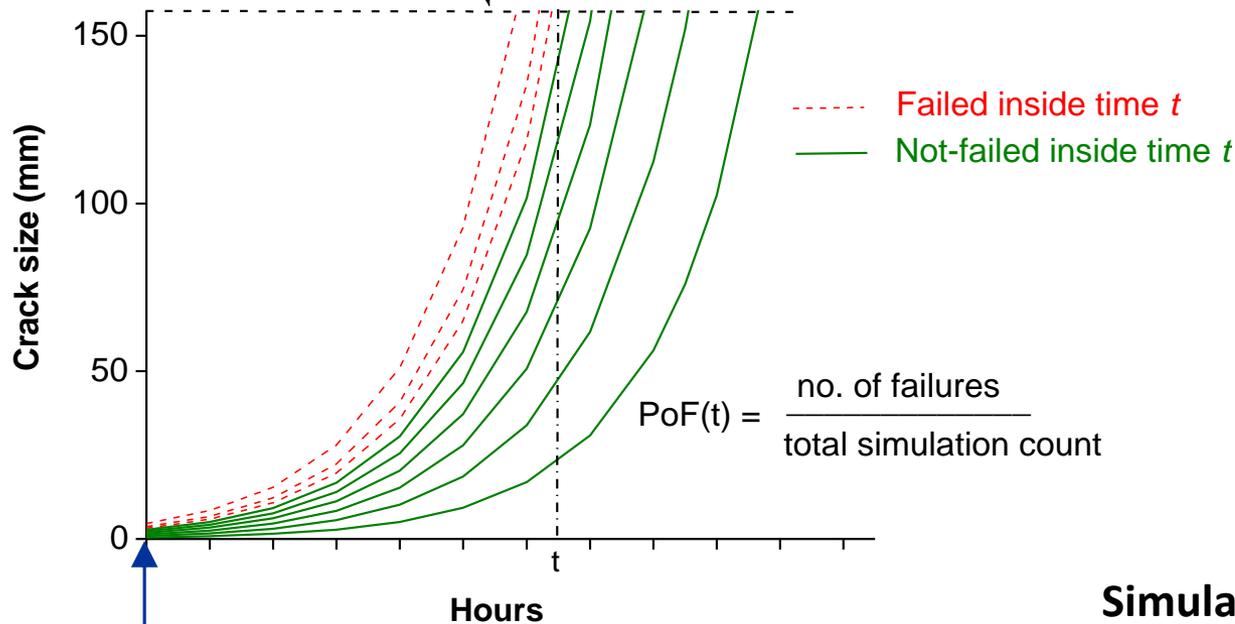
MSD Probabilistic Risk Analysis Methodology



Monte Carlo Simulation of the Probability of Failure

Calculation of the Probability of Failure

Limiting crack size = 157.48 mm

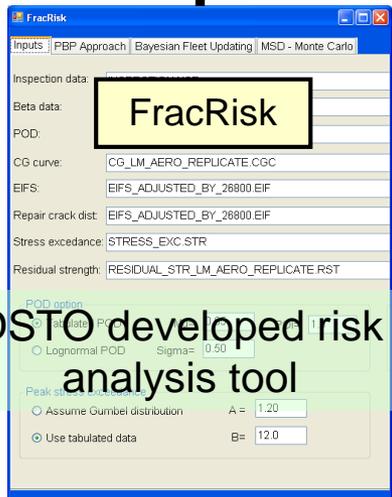


Randomly generated crack size at time zero (EIFS)

Probabilistic risk analysis of failure tool - FracRisk

Capabilities

Risk Curves

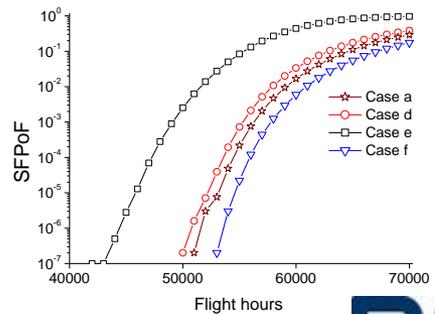
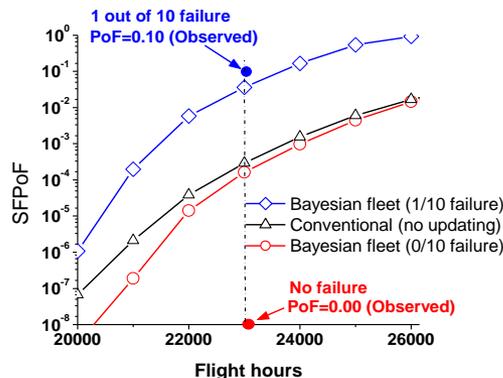
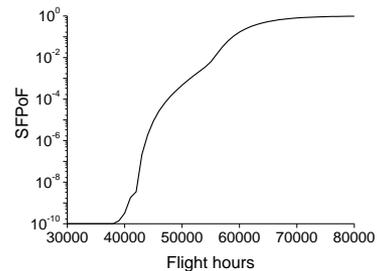


DSTO developed risk analysis tool

PROF method

Bayesian risk updating

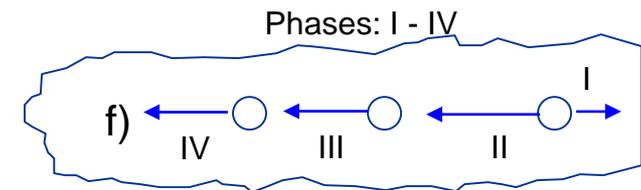
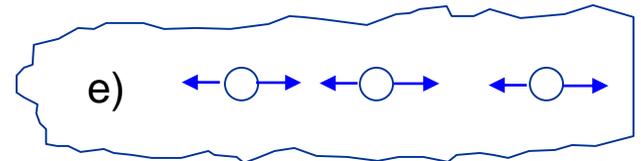
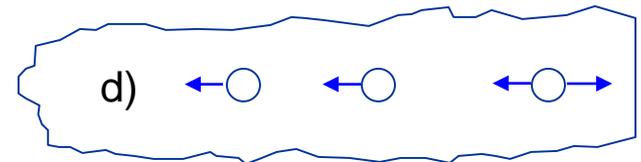
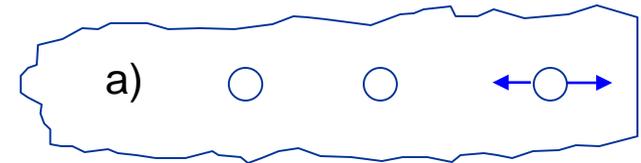
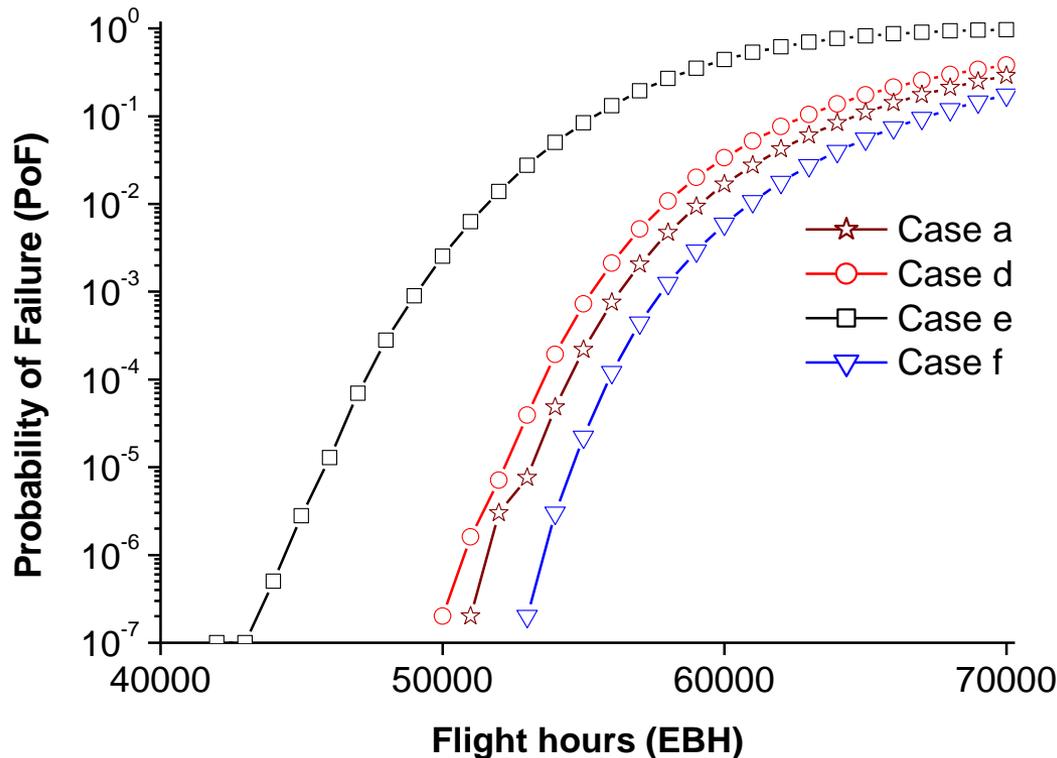
Multi-site damage (MSD) risk analysis



This module used in the analysis

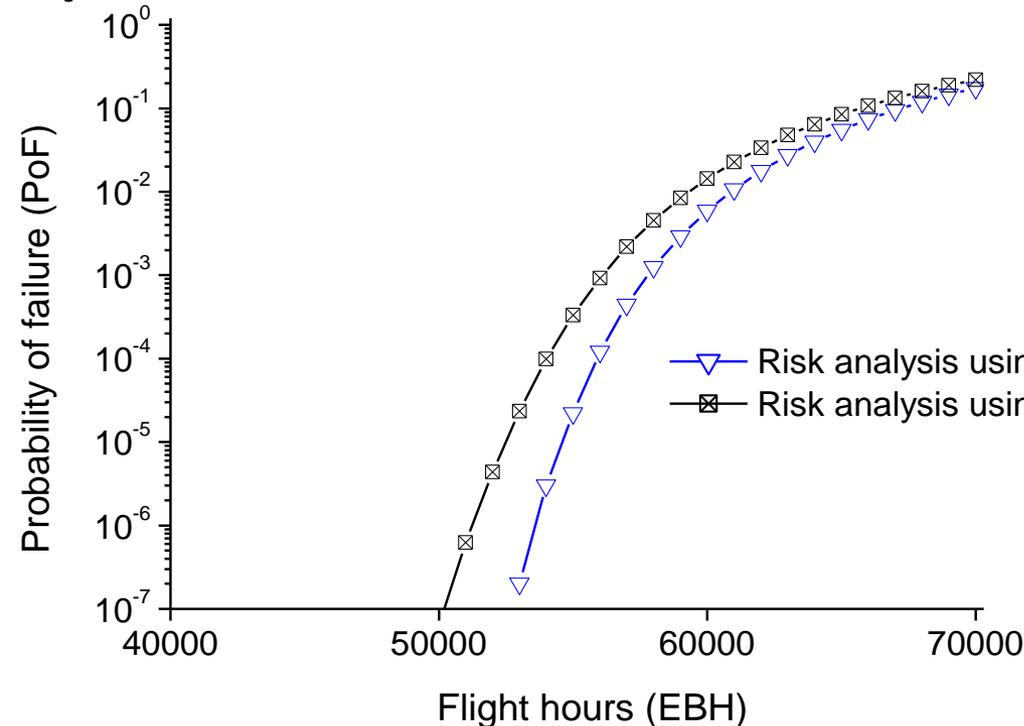
Probability of Failure (PoF) with increasing cracks numbers

 **Probability of failure increases with the increase of the number of cracks**



Probability of Failure (PoF) as a function of risk parameters

Probability of failure increases with the increase of unknown parameters

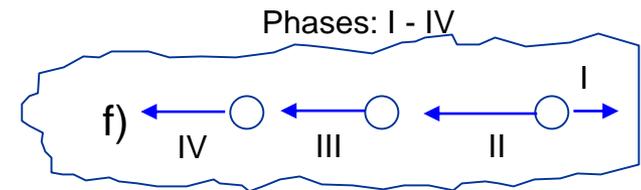


2 parameter risk analysis

1. Crack growth curve
2. Equivalent initial flaw size

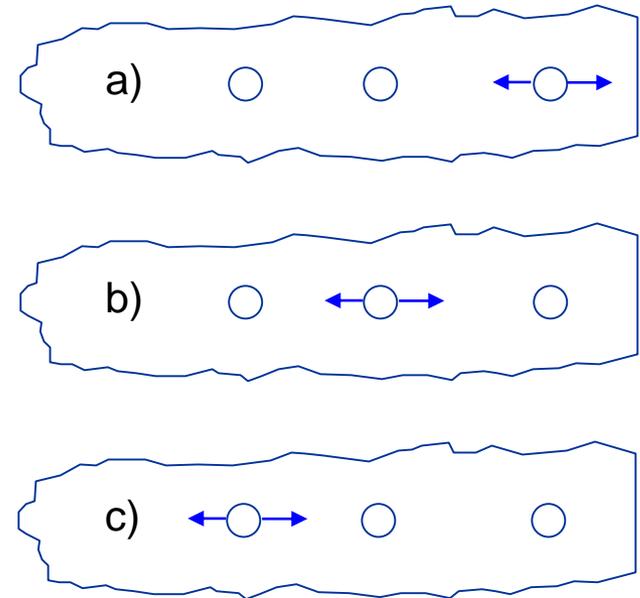
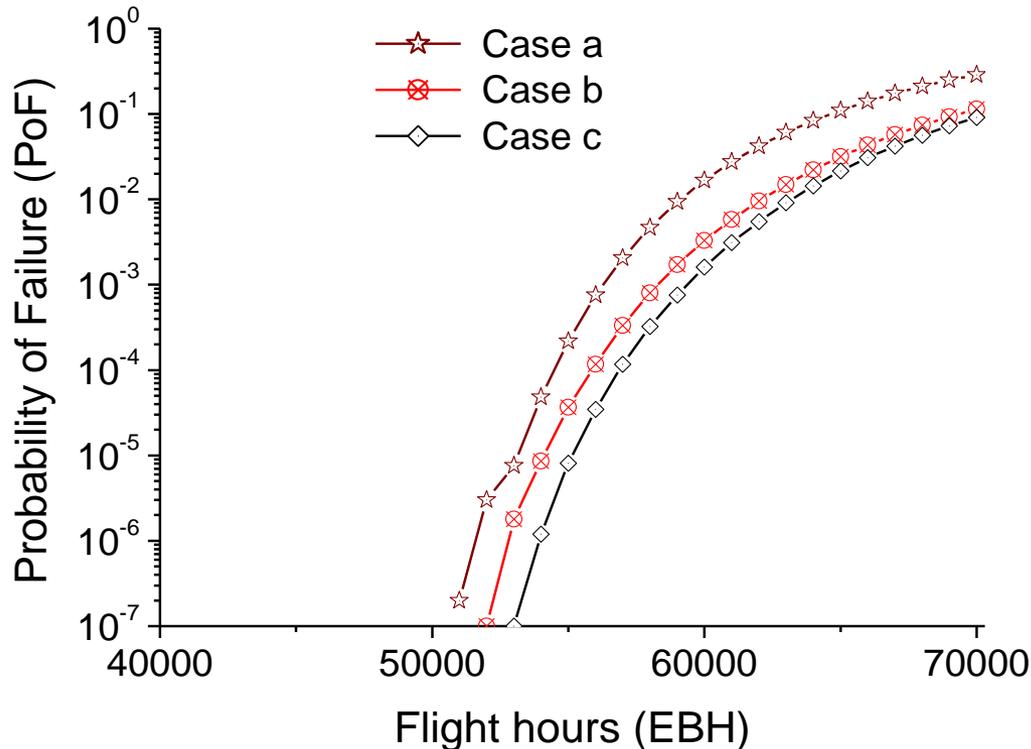
4 parameter risk analysis

1. Crack growth curve
2. Equivalent initial flaw size
3. Residual strength
4. Stress exceedance



Probability of Failure (PoF) by varying hole location

👉 **Probability of failure increases when the lead crack is closer to the edge**



Conclusions

- The presence of multi-site damage significantly increases the risk of failure for a structural component Updating risk analysis results can be done by utilising flight hours information
- The simplistic approach using only two sets of parameters to conduct risk analysis is shown to give lower risk values compared to the analysis using four parameters
- An increase in the number of cracks resulted to a corresponding increase in the risk of fracture
- Cracks closer to the edge of a component will result in a higher risk of failure due to its higher likelihood of becoming an edge crack



Questions ?